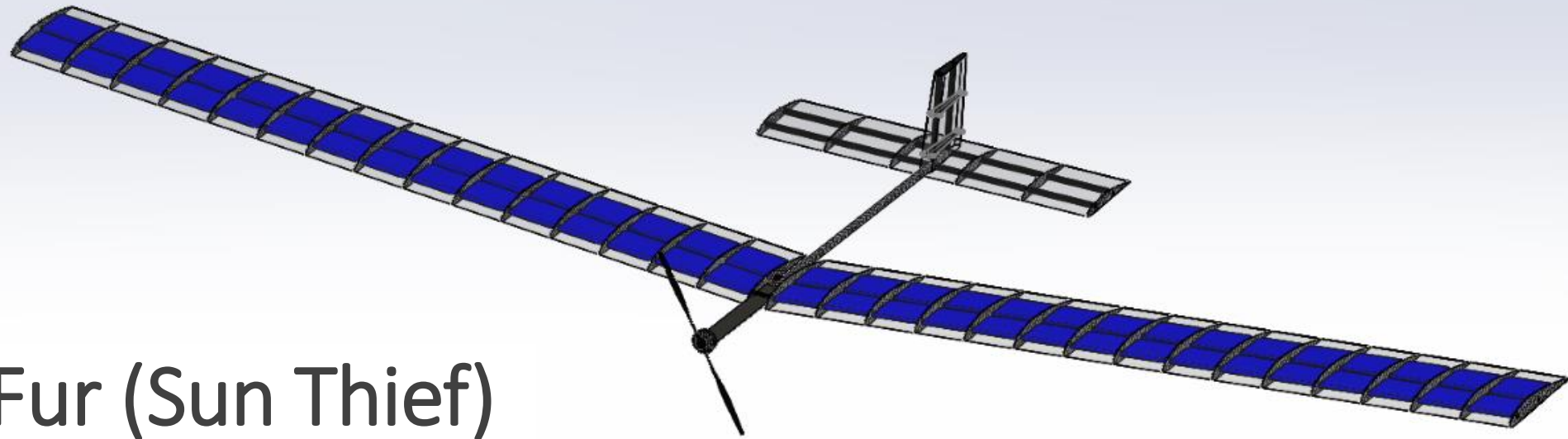




NORTHERN  
ARIZONA  
UNIVERSITY



Solis Fur (Sun Thief)

F1820 Solar Plane

# The Team



**Brandon Beaudoin**  
(Project Manager)



**Michael Broyles**  
(Website Designer)



**Nathan Zufelt**  
(Budget Manager)



**Ethan Smith**  
(Client Contact)

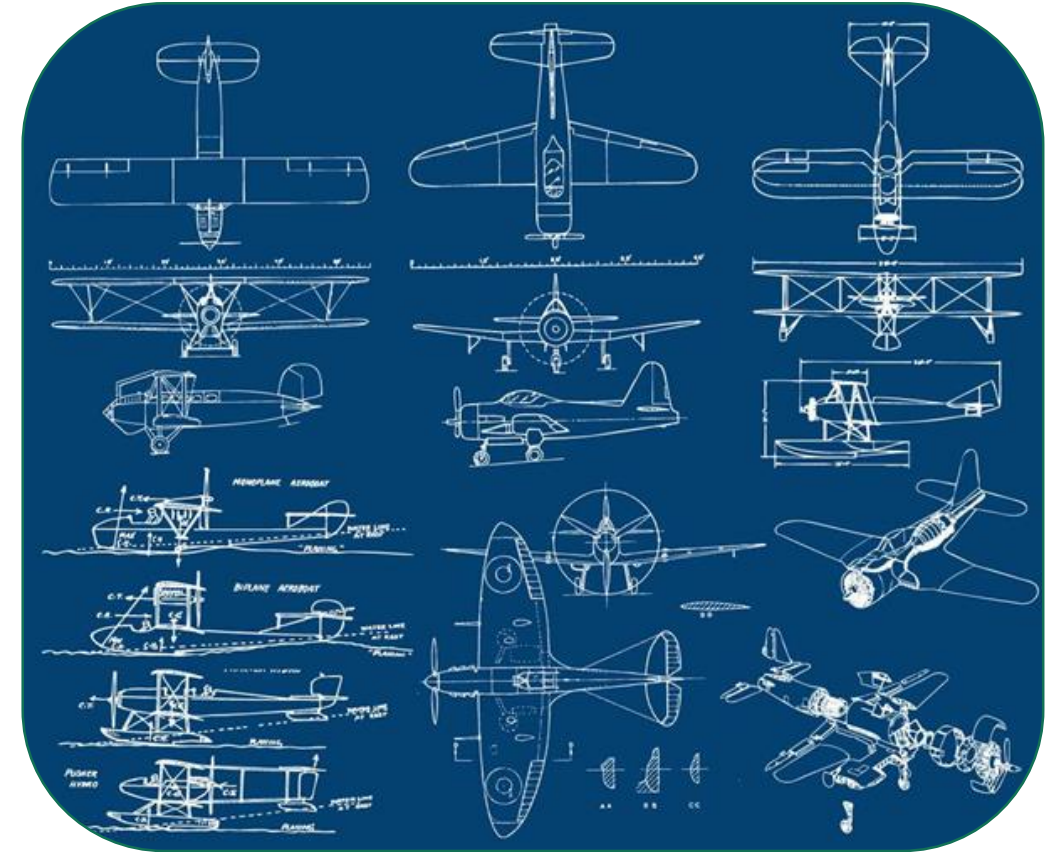


**Jonathan Hernandez**  
(Documentation Manager)



# Project Concept

Explore the use of engineering principles to design and build a solar powered RC aircraft capable of sustaining indefinite flight while the sun is out.



Plane Schematic [8]



# Project Sponsor / Customer



## David Trevas, PhD

- Provided customer requirements.
- Crucial input for design requirements.

## Sponsors

- Novakinetics AeroSystems
- Prometheus Solar
- Flagstaff Flyers

## Why is this important?

- Teaches students to use engineering principles in a real life application.
- Allows the use of renewable energy to power an RC plane.



# Project Review

## Individual Analysis Report Conclusions

### Control Surface Design:

- Control surface dimensioning used in conjunction with expected travel speeds to determine the force that is needed to actuate the surface.

### Airfoil Selection:

- A Rhodes32G-il airfoil was selected based on necessary lift and ease to manufacture. The chord length will be 350 mm to accommodate two rows of solar cells.

### Solar Power Input:

- Solar output for the panels will be higher than the standard testing conditions, resulting in a higher panel output than rated.

### Power Usage:

- Overall energy consumption of each component within electronics in the plane, the time of discharge rate of the battery.

### Wing Stress Analysis:

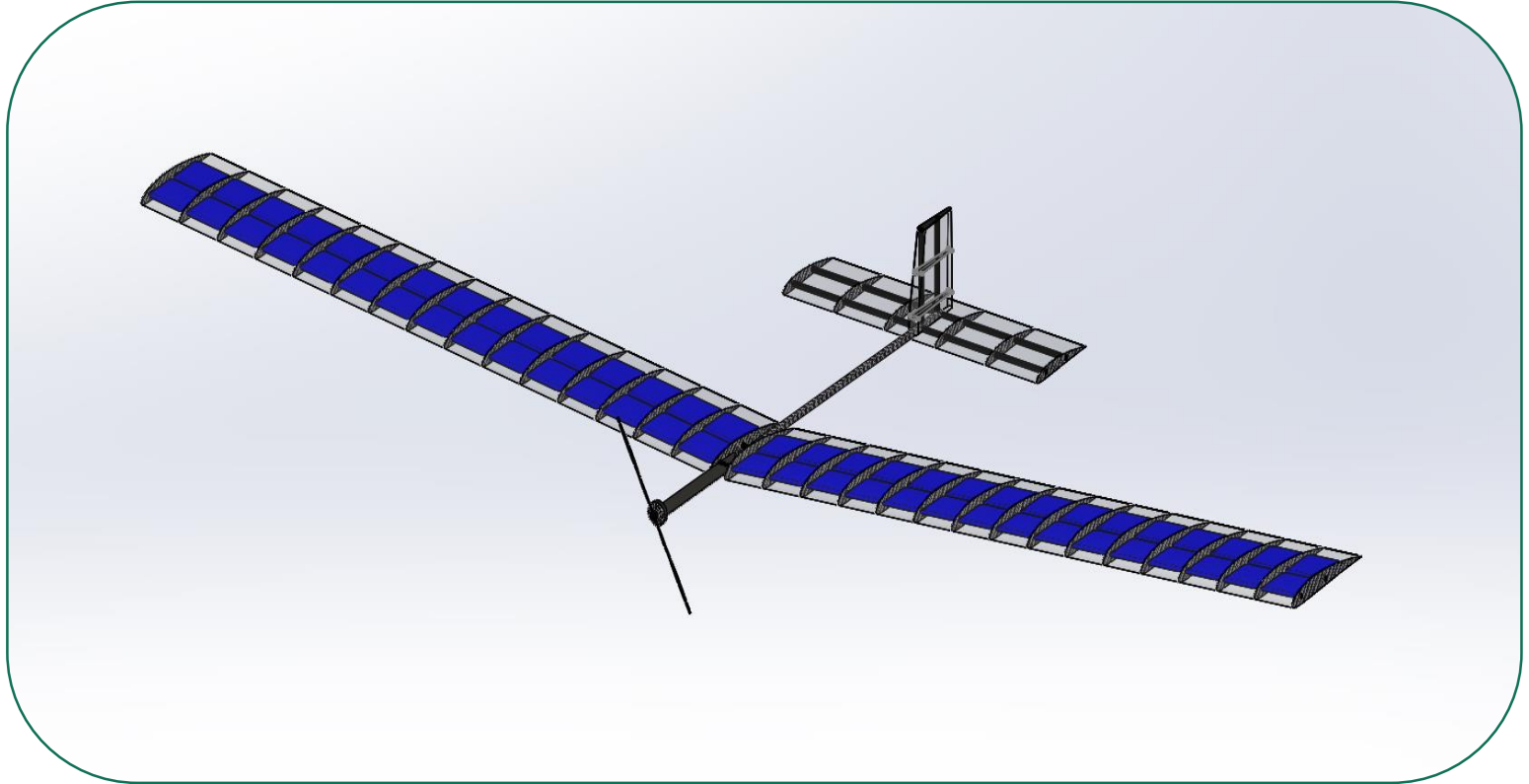
- Two 12mm Carbon Fiber spars provide the necessary strength to withstand the wing loading conditions.



# Proposed Design

## Specifications:

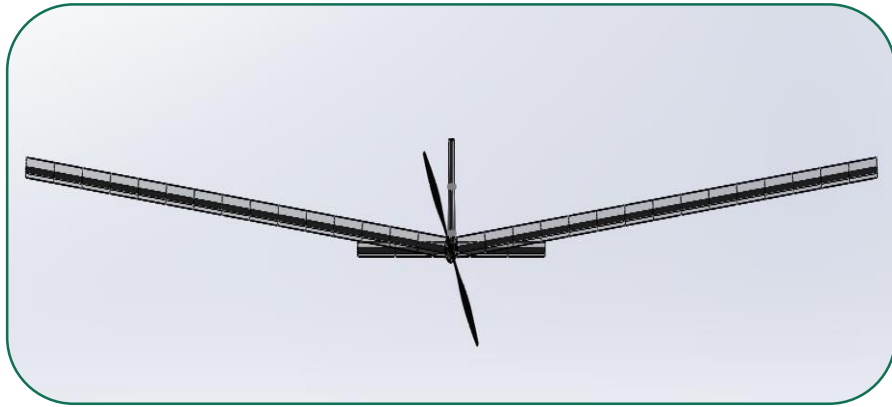
- Wing span: 13.25ft
- Weight: <8lbs
- Number of solar cells: 60
- Flight duration: Indefinite



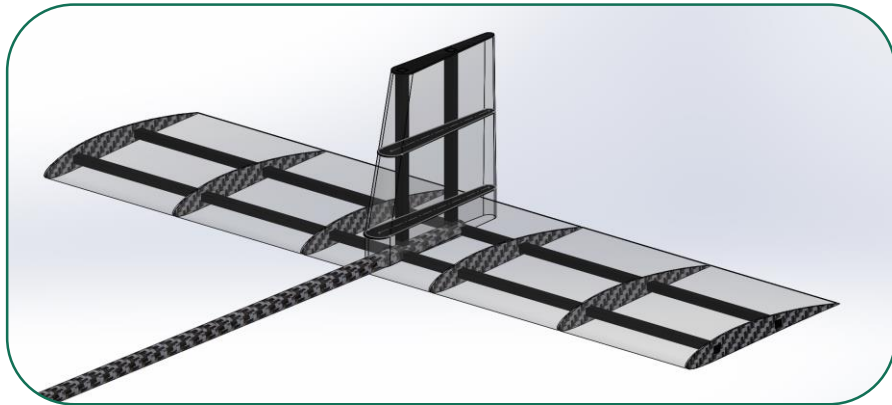
SolidWorks Rendering of Proposed Plane



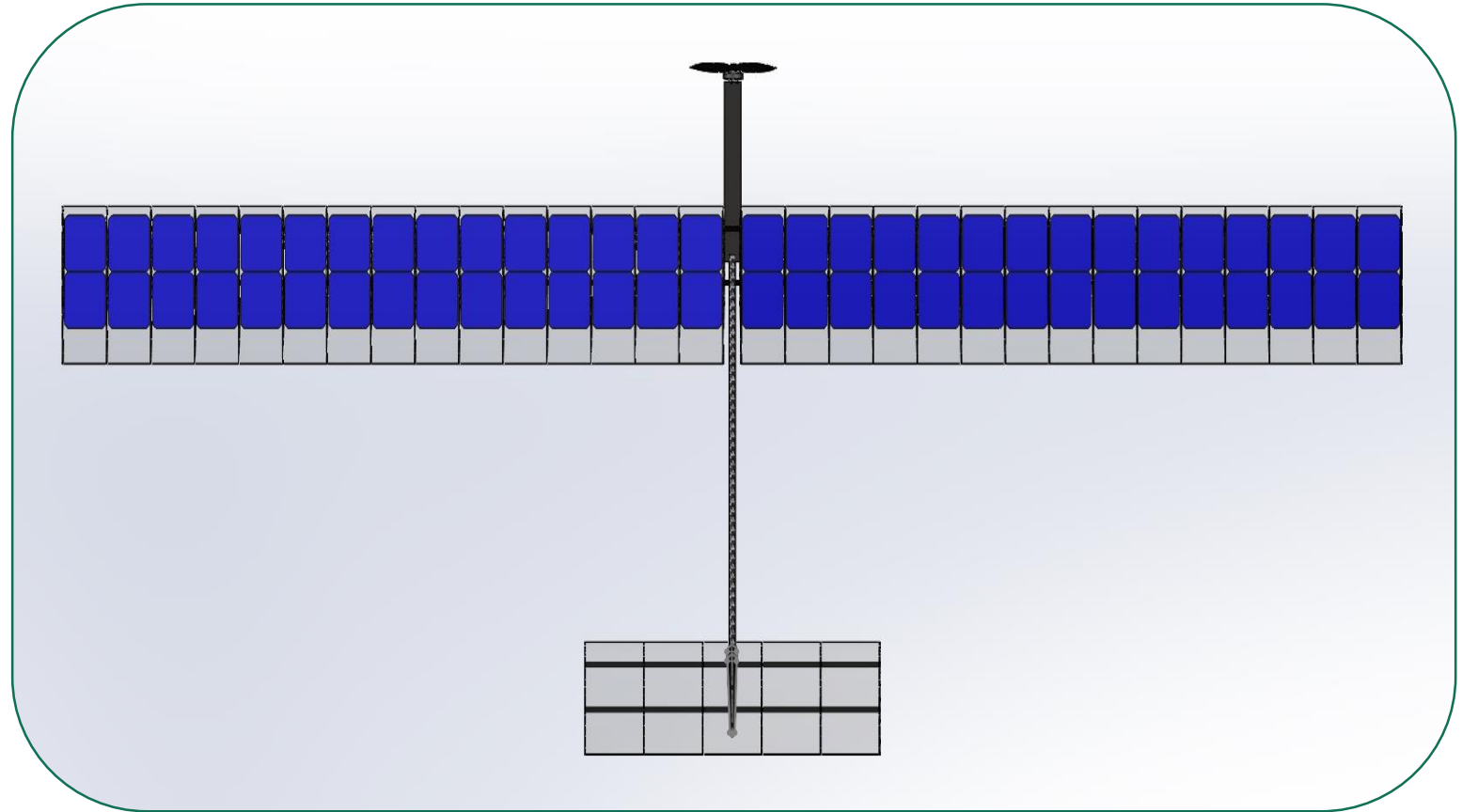
# Proposed Design



5 degree Dihedral



Proposed Tail



Top View

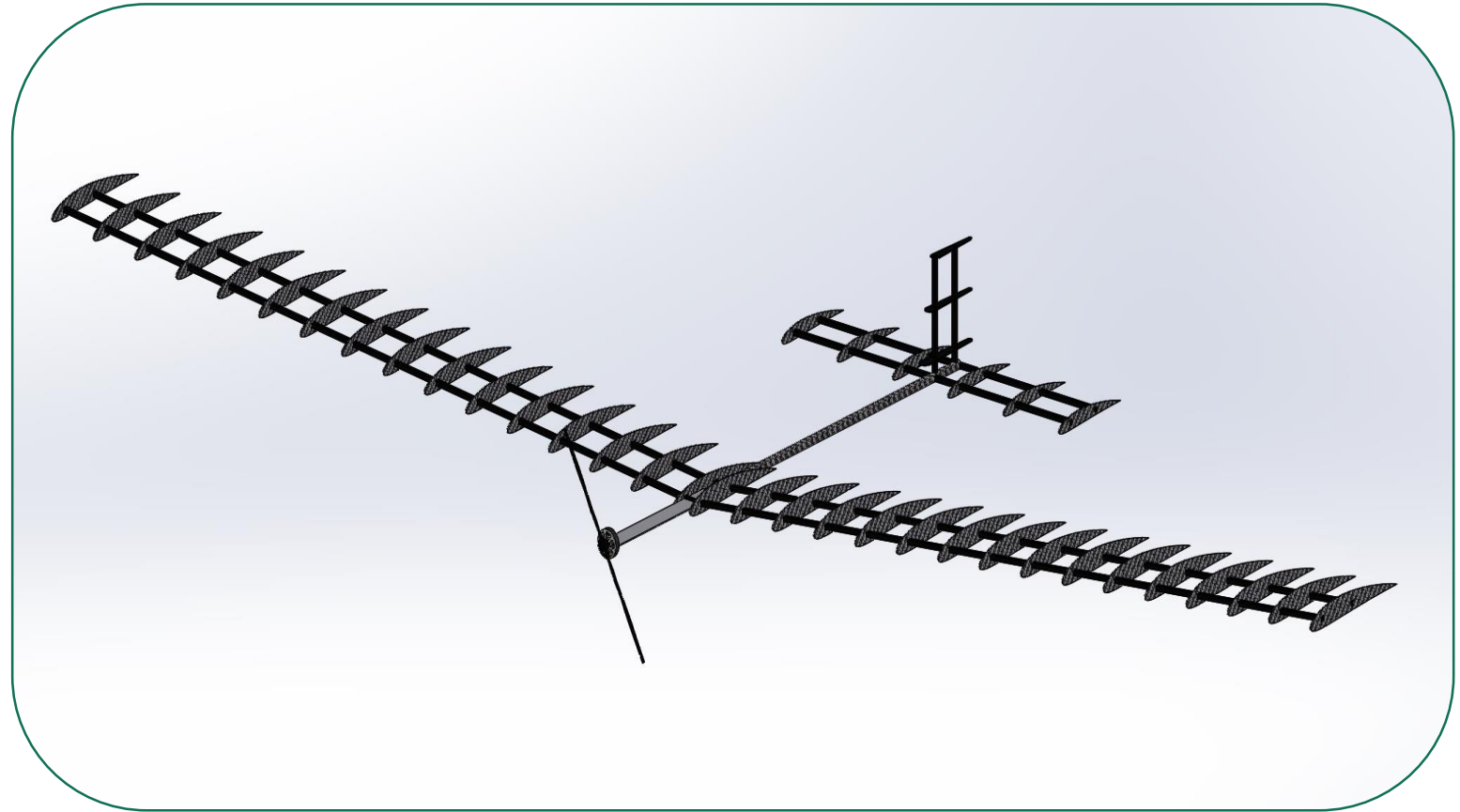
# Subsystems





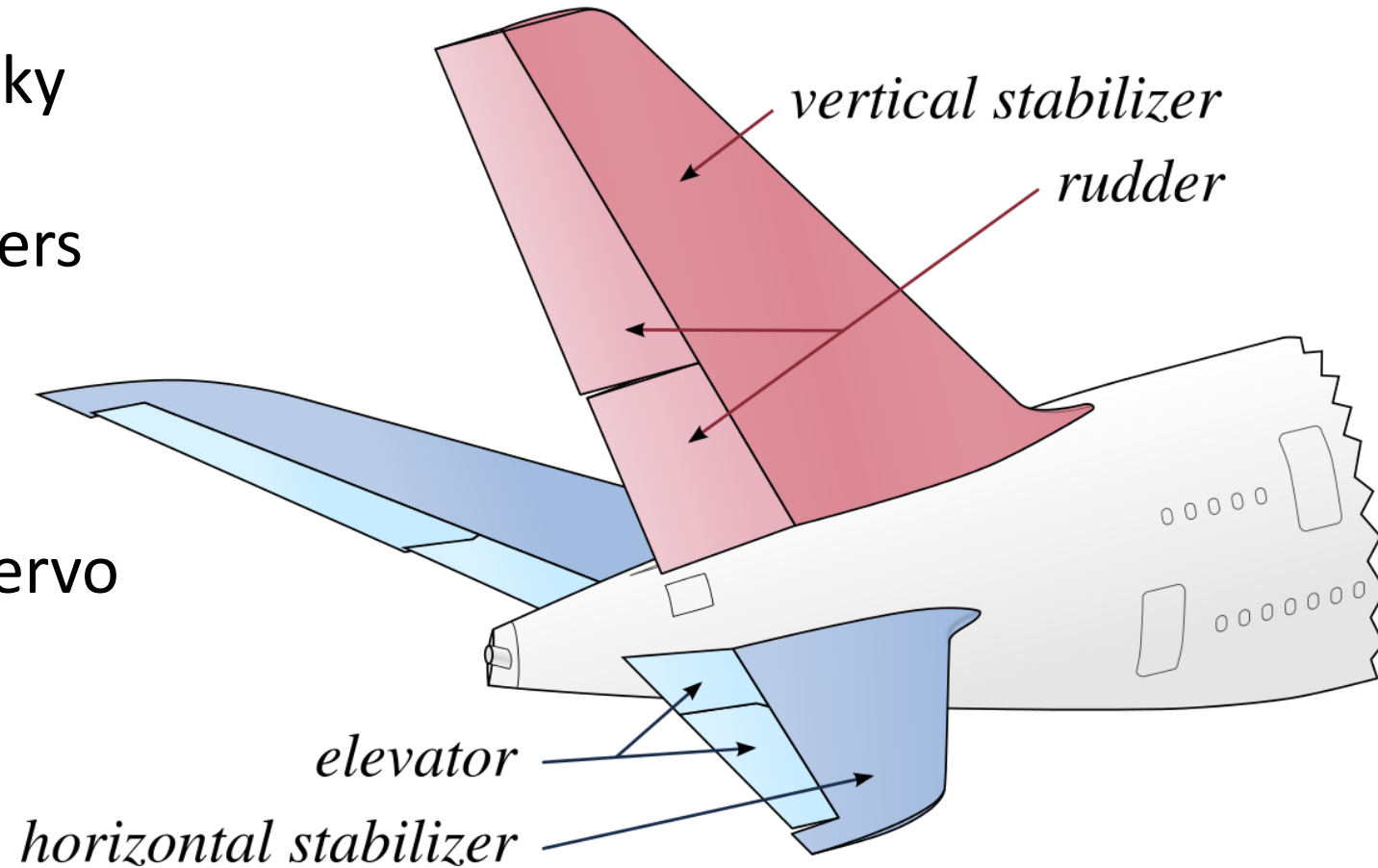
# Subsystems - Structure

- Simplified Plane Design
- Two spars per wing- 12mm round carbon fiber tubes
- Tail boom- 25 mm square carbon fiber tube
- Wing profile ribs- Balsa wood 350mm x 42mm



# Subsystems – Control Surfaces

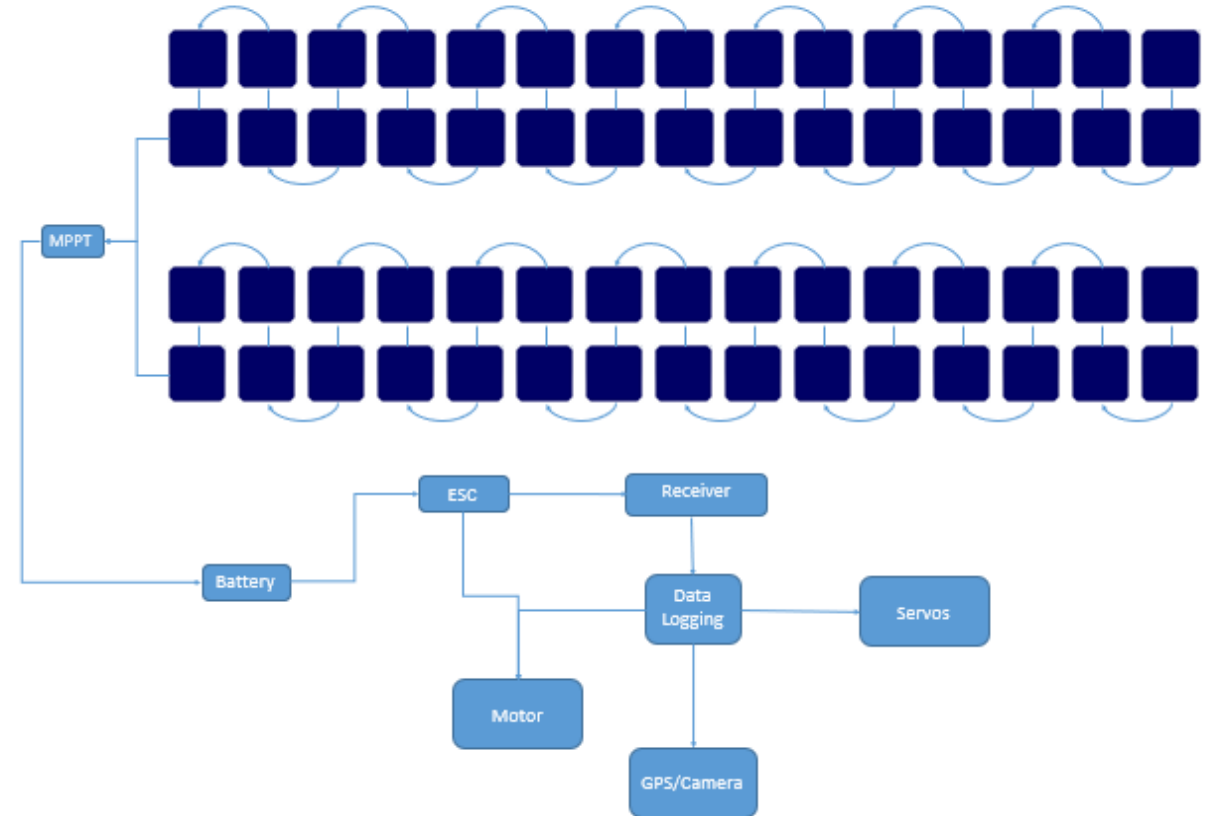
- Navigate the plane through the sky
- Work in conjunction with stabilizers
- Ensure safe flying conditions
- Sizing of control surface aids in servo selection



Tail Schematic [11]

# Subsystems – Electrical System

- Manage and distribute power where needed
- Communicate with the transmitter to respond to pilot input



# Subsystems – Solar Array

- Array of 30 in series per wing
  - 17.4 volts
- Each wing in parallel provides
  - 11.8 Amps
  - Each wing in parallel provides
- Harnesses energy from the sun
- Plane can endure indefinite flight



# Subsystems – Propulsion

- Provide thrust as a function of propeller size and voltage applied to the motor.
- Ensure that thrust available is higher than drag experienced by the entire plane.

**17X10 Folding Prop**



VOLTAGE V	NO LOAD		ON LOAD			LOAD TYPE						
	CURRENT A	SPEED rpm	CURRENT A	Pull g	Power W							
11.1	0.3	3996	1.0	200	11.1	CF15x5.5 Prop						
			2.5	400	27.8							
			4.3	580	47.7							
			11.1	0.3	3996	1.0	200	11.1	CF16x5.5 Prop			
						2.4	400	26.6				
						4.7	630	52.2				
						11.1	0.3	3996	1.0	200	11.1	CF17x5.5 Prop
									2.3	400	25.5	
									5.7	760	63.3	
14.8	0.3	5328	1.5	300	16.7	CF18x5.5 Prop						
			4.3	600	47.7							
			7.4	860	82.1							
			14.8	0.3	5328	1.4	300	20.7	CF15x5.5 Prop			
						3.5	600	51.8				
						6.5	910	96.2				
						14.8	0.3	5328	1.3	300	19.2	CF16x5.5 Prop
									3.5	600	51.8	
									7.1	990	105.1	
14.8	0.3	5328	1.8	400	26.6	CF17x5.5 Prop						
			4.7	800	69.6							
			8.5	1170	125.8							
			14.8	0.3	5328	1.8	400	26.6	CF18x5.5 Prop			
						5.1	800	75.5				
						10.8	1260	159.8				

**360KV Brushless Motor**



**Around 1100 grams of thrust**

# Design Requirements

## How Are We Meeting Design Requirements

### Indefinite Flight

- Glider style airfoil selected to optimize lift and reduce drag.
- Weight will be kept to a minimum by using balsa wood and carbon as the primary building materials.
- Low wing aspect ratio to reduce induced drag.

### Data Logging

- eLogger V4
  - Prop RPM (optical sensor)
  - Battery Voltage
  - Voltage and Current into battery
  - 50hz ---50 samples/s

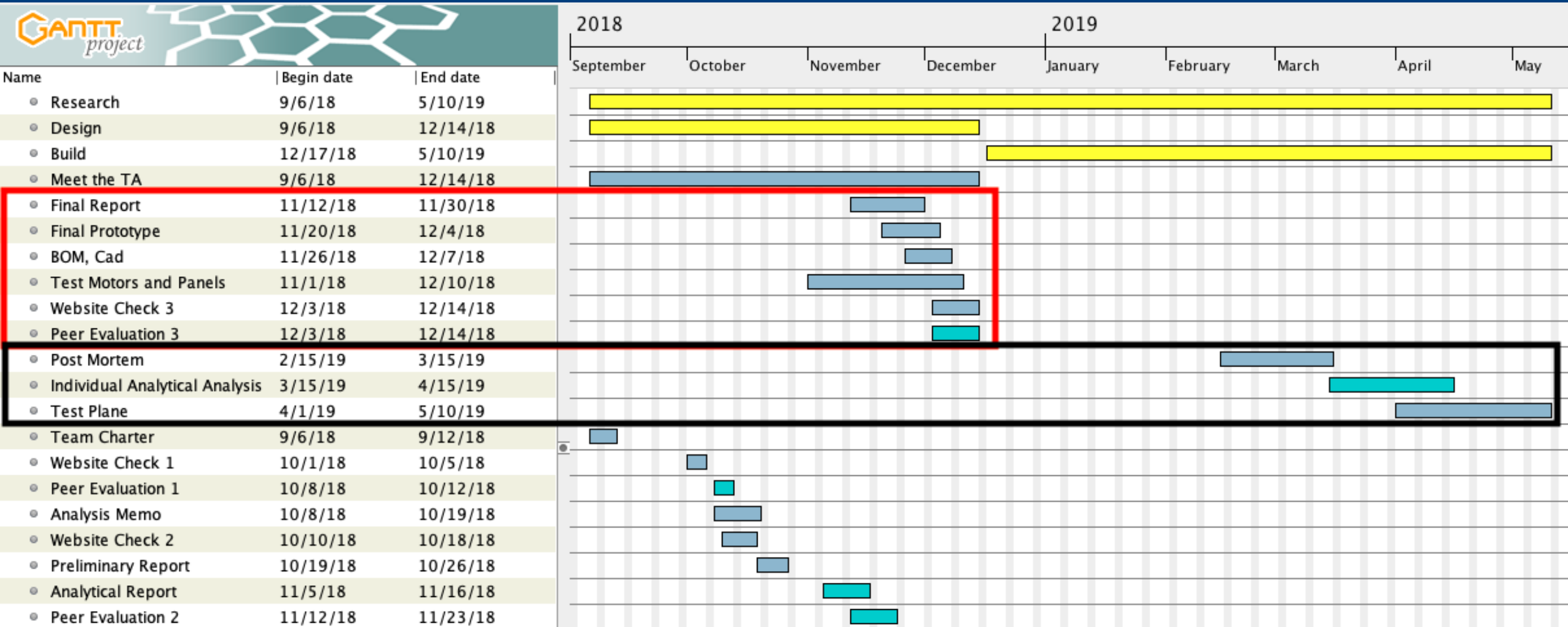
### FPV or Video

- GoPro (74g)
- RunCam 2 Action Camera (49g)





# Schedule





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# Questions?

